

NEET — Physics Formula Series

GRAVITATION

Complete Formula Sheet

All formulas, conditions, and key results at one place

G	g (surface)	Key Values
$6.67 \times 10^{-11} \text{ N m}^2 \text{ kg}^{-2}$	$9.8 \approx 10 \text{ m/s}^2$	$M_E = 6 \times 10^{24} \text{ kg}, \quad R_E = 6.4 \times 10^6 \text{ m}$

1. Newton's Law of Gravitation

$$\text{Gravitational Force: } F = \frac{Gm_1m_2}{r^2}$$

$$\text{Vector Form: } \vec{F}_{12} = -\frac{Gm_1m_2}{r^2} \hat{r}_{12} \quad (\text{attractive, along } \hat{r})$$

$$\text{Superposition: } \vec{F}_{\text{net}} = \vec{F}_{12} + \vec{F}_{13} + \vec{F}_{14} + \dots$$

★ **Important:** G is a universal constant. Gravitational force is always attractive, acts along the line joining the two masses, and obeys Newton's Third Law.

2. Acceleration Due to Gravity (g)

$$\text{On Surface: } g = \frac{GM}{R^2}$$

$$\text{At Latitude } \lambda : \quad g_\lambda = g - \omega^2 R \cos^2 \lambda$$

$$\text{At Height } h : \quad g_h = \frac{GM}{(R+h)^2} = g \left(1 + \frac{h}{R}\right)^{-2} \approx g \left(1 - \frac{2h}{R}\right) \quad (\text{At Equator } h \ll R): \quad g_{\text{eq}} = g - \omega^2 R$$

$$\text{At Depth } d : \quad g_d = g \left(1 - \frac{d}{R}\right) \Rightarrow g_d = 0 \text{ at centre}$$

$$\text{At Poles (max): } g_{\text{pole}} = g \quad (\text{no rotation effect})$$

★ **Important:** g is maximum at poles and minimum at equator. $g = 0$ at the centre of the Earth. g decreases both above and below the surface, but the rate of decrease is different.

3. Gravitational Field Intensity (\vec{g} or \vec{E})

$$\text{Outside Solid Sphere } (r > R) : \quad g = \frac{GM}{r^2}$$

$$\text{Definition: } \vec{g} = \frac{\vec{F}}{m} \quad (\text{force per unit test mass})$$

$$\text{Due to Point Mass } M : \quad g = \frac{GM}{r^2} \quad (\text{towards } M)$$

$$\text{Inside Solid Sphere } (r < R) : \quad g = \frac{GMr}{R^3}$$

$$\begin{aligned} \text{Inside Hollow Shell: } g &= 0 \\ \text{Outside Hollow Shell: } g &= \frac{GM}{r^2} \end{aligned}$$

$$\text{Relation with Potential: } \vec{g} = -\frac{dV}{dr} \Rightarrow \vec{g} = -\nabla V$$

4. Gravitational Potential (V)

$$\text{Definition: } V = \frac{W}{m} = -\int_{\infty}^r \vec{g} \cdot d\vec{r} \quad (\text{always negative})$$

$$\text{At Centre of Solid Sphere: } V_c = -\frac{3GM}{2R} = \frac{3}{2}V_{\text{surface}}$$

$$\text{Due to Point Mass: } V = -\frac{GM}{r}$$

$$\text{Inside/On Hollow Shell: } V = -\frac{GM}{R} \quad (\text{constant everywhere})$$

$$\text{Outside Solid Sphere } (r > R) : V = -\frac{GM}{r}$$

$$\text{Inside Solid Sphere } (r < R) : V = -\frac{GM}{2R^3}(3R^2 - r^2)$$

$$\text{Outside Hollow Shell: } V = -\frac{GM}{r}$$

★ **Important:** Gravitational potential is always negative (taking $V = 0$ at infinity). The potential inside a hollow shell is constant and equal to the surface value.

5. Gravitational Potential Energy (U)

$$\text{Two Point Masses: } U = -\frac{Gm_1m_2}{r}$$

$$\text{At Height } h : U_h = -\frac{GMm}{R+h}$$

$$\text{Mass } m \text{ near Earth: } U = -\frac{GMm}{r}$$

$$\text{Change in PE } (h \ll R) : \Delta U = U_h - U_s = mgh$$

$$\text{On Earth's Surface: } U_s = -\frac{GMm}{R} = -mgR$$

$$\text{System of } n \text{ masses: } U = -G \sum_{i < j} \frac{m_i m_j}{r_{ij}}$$

6. Escape Velocity (v_e)

$$\text{From Surface: } v_e = \sqrt{\frac{2GM}{R}} = \sqrt{2gR}$$

$$\text{At Height } h : v_e(h) = \sqrt{\frac{2GM}{R+h}}$$

$$\text{Numerical: } v_e \approx 11.2 \text{ km/s (Earth)}$$

$$\text{Relation: } v_e = \sqrt{2} v_o \Rightarrow v_e : v_o = \sqrt{2} : 1$$

$$\text{In terms of } g_h : v_e = \sqrt{2g_h(R+h)}$$

★ **Important:** Escape velocity is independent of the mass and direction of projection. It depends only on the mass and radius of the planet.

7. Orbital Motion of Satellites

$$\text{Orbital Velocity: } v_o = \sqrt{\frac{GM}{r}} = \sqrt{\frac{GM}{R+h}}$$

$$\text{In terms of } g, R : v_o = R \sqrt{\frac{g}{R+h}}$$

$$\text{Near Surface } (h \approx 0) : v_o = \sqrt{gR} \approx 7.9 \text{ km/s}$$

$$\text{Time Period: } T = \frac{2\pi r}{v_o} = 2\pi\sqrt{\frac{r^3}{GM}} = 2\pi\sqrt{\frac{(R+h)^3}{GM}}$$

$$\text{Near Surface: } T_{\min} = 2\pi\sqrt{\frac{R}{g}} \approx 84 \text{ min}$$

$$\text{Height from Period: } h = \left(\frac{GMT^2}{4\pi^2}\right)^{1/3} - R$$

8. Energy of a Satellite in Circular Orbit

$$\begin{aligned} \text{KE: } KE &= \frac{1}{2}mv_o^2 = \frac{GMm}{2r} \quad (\text{always positive}) & \text{Total Energy: } E = KE + PE &= -\frac{GMm}{2r} \quad (\text{always negative}) \\ \text{PE: } PE &= -\frac{GMm}{r} \quad (\text{always negative}) & \text{Binding Energy: } BE &= -E = \frac{GMm}{2r} \\ \text{Key Ratio: } & KE = -E; \quad PE = 2E; \quad PE = -2KE \end{aligned}$$

★ **Important:** Total energy of a satellite is always negative, indicating a bound system. The magnitude of KE equals the Binding Energy. As orbit radius increases, KE decreases but $|E|$ also decreases, so total energy increases (becomes less negative).

9. Kepler's Laws of Planetary Motion

First Law (Orbit): Each planet moves in an elliptical orbit with the Sun at one focus.

Second Law (Area): $\frac{dA}{dt} = \frac{L}{2m} = \text{constant}$ (Equal areas swept in equal times \Rightarrow Angular momentum is conserved)

Third Law ($T^2 \propto a^3$):

$$T^2 = \frac{4\pi^2}{GM}a^3 \Rightarrow \frac{T_1^2}{T_2^2} = \frac{a_1^3}{a_2^3}$$

$$\text{For Circular Orbit } (a = r) : T^2 = \frac{4\pi^2}{GM}r^3$$

Velocity at Perihelion: $v_p \cdot r_p = v_a \cdot r_a$ (conservation of L) Semi-major axis: $a = \frac{r_p + r_a}{2}$

10. Geostationary and Polar Satellites

Geostationary Orbit: Time period $T = 24$ hours, orbits in equatorial plane, same direction as Earth's rotation.

Height of Geostationary: $h \approx 36,000 \text{ km} = 3.6 \times 10^7 \text{ m}$

Radius of Orbit: $r_g \approx 42,400 \text{ km}$

Polar Satellite: Orbits at $\approx 500\text{--}800 \text{ km}$ height, $T \approx 100 \text{ min}$, passes over both poles — used for remote sensing.

11. Weightlessness

Condition: Apparent weight $= m(g - a) = 0$ when $a = g$ (free fall condition).

In Satellite: Both satellite and astronaut are in free fall with same acceleration $= \frac{GM}{r^2}$ — hence weightlessness.

At Earth's Centre: $g = 0$ at centre \Rightarrow weight $= 0$.

NOT the same as: Zero gravity — gravity still acts, but the normal reaction from the surface becomes zero.

12. Quick Comparison — Solid Sphere vs Hollow Shell

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shaderow g inside	$\frac{GMr}{R^3}$ (varies with r)	0 (zero everywhere)
g outside	$\frac{GM}{r^2}$	$\frac{GM}{r^2}$
shaderow V inside	$-\frac{GM}{2R^3}(3R^2 - r^2)$ (varies)	$-\frac{GM}{R}$ (constant)
V outside	$-\frac{GM}{r}$	$-\frac{GM}{r}$
shaderow V at centre	$-\frac{3GM}{2R}$	$-\frac{GM}{R}$
V at surface	$-\frac{GM}{R}$	$-\frac{GM}{R}$

13. Master Formula List at a Glance

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shaderow 1	Gravitational Force	$F = \frac{Gm_1m_2}{r^2}$
2	g on Surface	$g = \frac{GM}{R^2}$
shaderow 3	g at Height h	$g_h \approx g \left(1 - \frac{2h}{R}\right)$
4	g at Depth d	$g_d = g \left(1 - \frac{d}{R}\right)$
shaderow 5	g at Latitude λ	$g_\lambda = g - \omega^2 R \cos^2 \lambda$
6	Gravitational Potential	$V = -\frac{GM}{r}$
shaderow 7	Gravitational PE	$U = -\frac{GMm}{r}$
8	Escape Velocity	$v_e = \sqrt{\frac{2GM}{R}} = \sqrt{2gR}$
shaderow 9	Orbital Velocity	$v_o = \sqrt{\frac{GM}{r}} = \sqrt{\frac{gR^2}{R+h}}$
10	Time Period	$T = 2\pi \sqrt{\frac{r^3}{GM}}$
shaderow 11	KE of Satellite	$KE = \frac{GMm}{2r}$
12	PE of Satellite	$PE = -\frac{GMm}{r}$
shaderow 13	Total Energy	$E = -\frac{GMm}{2r}$
14	Binding Energy	$BE = \frac{GMm}{2r}$

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shaderow 15	Kepler's Third Law	$T^2 = \frac{4\pi^2}{GM}a^3$
16	Areal Velocity	$\frac{dA}{dt} = \frac{L}{2m} = \text{const}$
shaderow 17	v_e vs v_o Relation	$v_e = \sqrt{2} v_o$
18	V inside Solid Sphere	$V = -\frac{GM}{2R^3}(3R^2 - r^2)$
shaderow 19	V at Centre	$V_c = -\frac{3GM}{2R}$
20	Gravitational Field	$\vec{g} = -\frac{dV}{dr}\hat{r}$